

Conservative Finite Volume Schemes for Meteorological Applications

Rupert Klein

Mathematik & Informatik, Freie Universität Berlin
Data & Computation, Potsdam Institute for Climate Impact Research

Emmanuel Audusse	(Paris 13)
Ann Almgren	(Lawrence Berkeley Lab.)
Nicola Botta	(PIK)
Michael Minion	(UNC-Chapel Hill)
Claus-Dieter Munz	(Uni-Stuttgart)
Antony Owinoh	(FUB / PIK)
Stefan Vater	(FUB / PIK)

Thanks to ...

Motivation

Balancing gravity

Momentum conservation vs. Coriolis

A low-Mach / low-Froude conservative scheme

low Mach number Godunov-Type scheme

robust, exact projection

low-Froude extension

$$\mathbf{u}_t + \mathbf{u} \cdot \nabla \mathbf{u} + w \mathbf{u}_z + \nabla \pi = \mathbf{S}_u$$

$$w_t + \mathbf{u} \cdot \nabla w + w w_z + \pi_z = -\theta' + S_w$$

$$\theta'_t + \mathbf{u} \cdot \nabla \theta' + w \theta'_z = S'_\theta$$

$$\nabla \cdot (\rho_0 \mathbf{u}) + (\rho_0 w)_z = 0$$

$$\theta = 1 + \varepsilon^4 \theta'(\mathbf{x}, z, t) + o(\varepsilon^4)$$

Anelastic Boussinesque Model

10 km / 20 min

$$(\partial_\tau + \mathbf{u}^{(0)} \cdot \nabla) q = 0$$

$$q = \zeta^{(0)} + \Omega_0 \beta \eta + \frac{\Omega_0}{\rho^{(0)}} \frac{\partial}{\partial z} \left(\frac{\rho^{(0)}}{d\Theta/dz} \theta^{(3)} \right)$$

$$\zeta^{(0)} = \nabla^2 \pi^{(3)}, \quad \theta^{(3)} = -\frac{\partial \pi^{(3)}}{\partial z}, \quad \mathbf{u}^{(0)} = \frac{1}{\Omega} \mathbf{k} \times \nabla \pi^{(3)}$$

Quasi-geostrophic theory

1000 km / 2 days

$$\frac{\partial Q_T}{\partial t} + \nabla \cdot \mathbf{F}_T = S_T$$

$$\frac{\partial Q_q}{\partial t} + \nabla \cdot \mathbf{F}_q = S_q$$

$$Q_\varphi = \int_{z_a}^{H_a} \rho \varphi dz, \quad \mathbf{F}_\varphi = \int_{z_a}^{H_a} \rho (\mathbf{u} \varphi + (\widehat{\mathbf{u}} \varphi') + \mathbf{D}^\varphi) dz, \quad (\varphi \in \{T, q\})$$

$$T = T_s(t, \mathbf{x}) + \Gamma(t, \mathbf{x}) \left(\min(z, H_T) - z_a \right), \quad q = q_s(t, \mathbf{x}) \exp \left(-\frac{z - z_a}{H_q} \right)$$

$$\rho = \rho_s \exp \left(-\frac{z}{h_w} \right), \quad p = p_s \exp \left(-\frac{\gamma z}{h_w} \right) + p_0(t, \mathbf{x}) + g \rho_s \int_0^z T_s dz'$$

$$\mathbf{u} = \mathbf{u}_g + \mathbf{u}_a, \quad f \rho_s \mathbf{k} \times \mathbf{u}_g = -\nabla_x p, \quad \mathbf{u}_a = \alpha \nabla p_0$$

V. Petoukhov et al., *CLIMBER-2 ...*, *Climate Dynamics*, 16, (2000)

EMIC - equations (CLIMBER-2)

10000 km / 1 season

Scales in the Atmosphere

Three-dimensional compressible flow equations

$$\rho_t + \nabla \cdot (\rho \mathbf{v}) = 0$$

$$(\rho \mathbf{v})_t + \nabla \cdot (\rho \mathbf{v} \circ \mathbf{v}) + \frac{1}{\mathbf{M}^2} \nabla p + \frac{1}{\mathbf{Ro}} \boldsymbol{\Omega} \times \rho \mathbf{v} = \mathcal{S}_{\rho \mathbf{v}} - \frac{1}{\mathbf{Fr}^2} \rho g \mathbf{k}$$

$$(\rho e)_t + \nabla \cdot (\mathbf{v} [\rho e + p]) = S_{\rho e}$$

$$(\rho Y_j)_t + \nabla \cdot (\rho Y_j \mathbf{v}) = S_{\rho Y_j}$$

$$(\rho e) = \frac{p}{\gamma - 1} + \frac{1}{2} \rho \mathbf{v}^2 + \rho \sum_{j=1}^N Q_j Y_j$$

Many asymptotic regimes — a single numerical scheme?

The “Truth”

Why conservation?

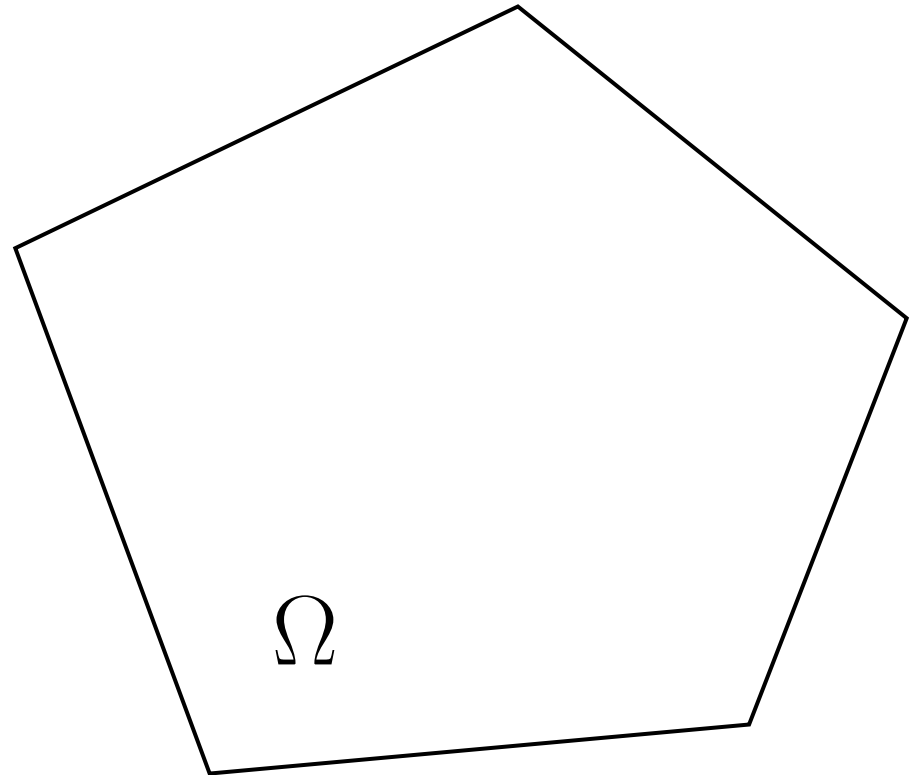
The integral conservation law

$$\int_{\Omega} \Phi \, d\mathbf{x} \Big|_{t_1}^{t_2} = - \int_{t_1}^{t_2} \oint_{\partial\Omega} \mathbf{f}_{\Phi} \cdot \mathbf{n} \, d\sigma \, dt$$

is uniformly valid for **arbitrary** control volumes



Mathematically sound subgrid closures



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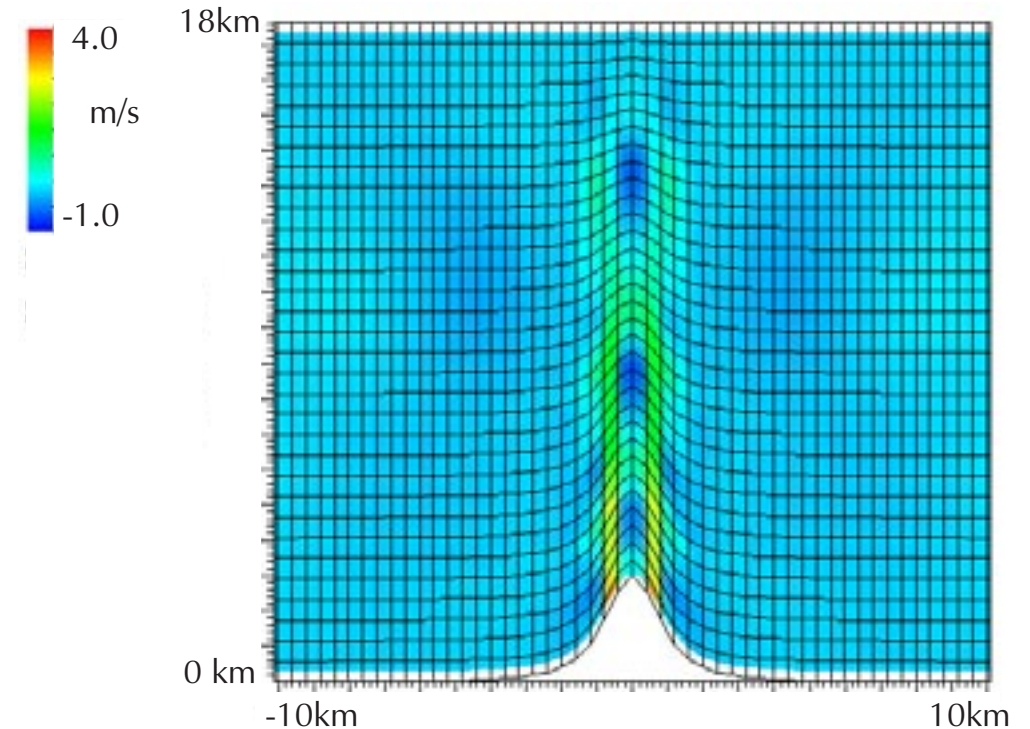
robust, exact projection

low-Froude extension

Spurious winds over steep orography

- atmosphere at rest
- 3000 m mountain
- 3D compressible inviscid flow eqs.
- standard finite volume scheme
- 128×32 grid cells
- velocities after about 60 min

Various finite difference / finite volume schemes produce comparable results.



Accumulation of unbalanced truncation errors

- The background pressure / temperature structure of the atmosphere is (almost) everywhere dominated by hydrostatic balance.

- Mathematically, this follows from the distinguished limits

$$\mathbf{Fr}^2 / \mathbf{M}^2 = O(1), \text{Ro} \gg \mathbf{M}^2, (\mathbf{M} \rightarrow 0)$$

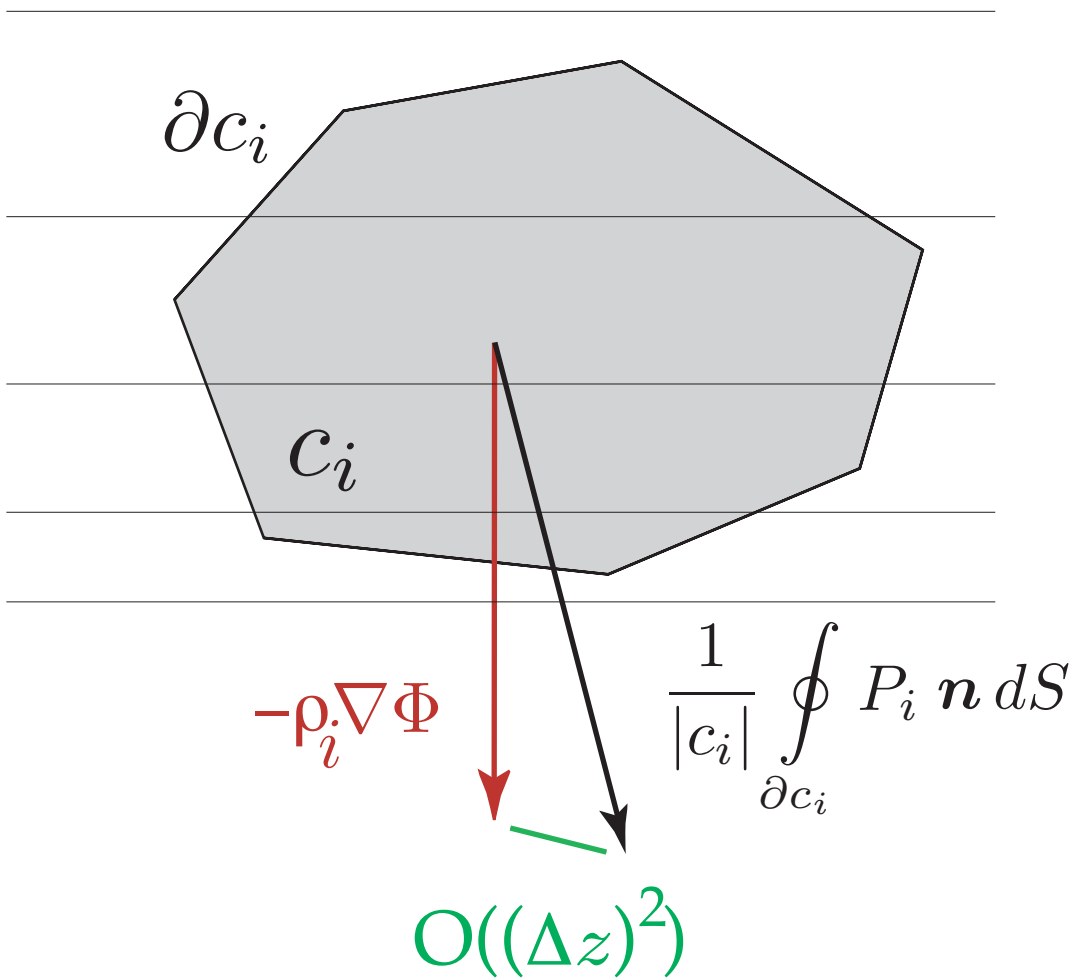
- Vertical momentum balance

$$(\rho w)_t + \nabla \cdot (\rho \mathbf{v} w) + \frac{1}{\text{Ro}} \mathbf{k} \cdot (\boldsymbol{\Omega} \times \mathbf{v}) = - \left(\frac{1}{\mathbf{M}^2} \frac{\partial p}{\partial z} + \frac{1}{\mathbf{Fr}^2} \rho \right).$$

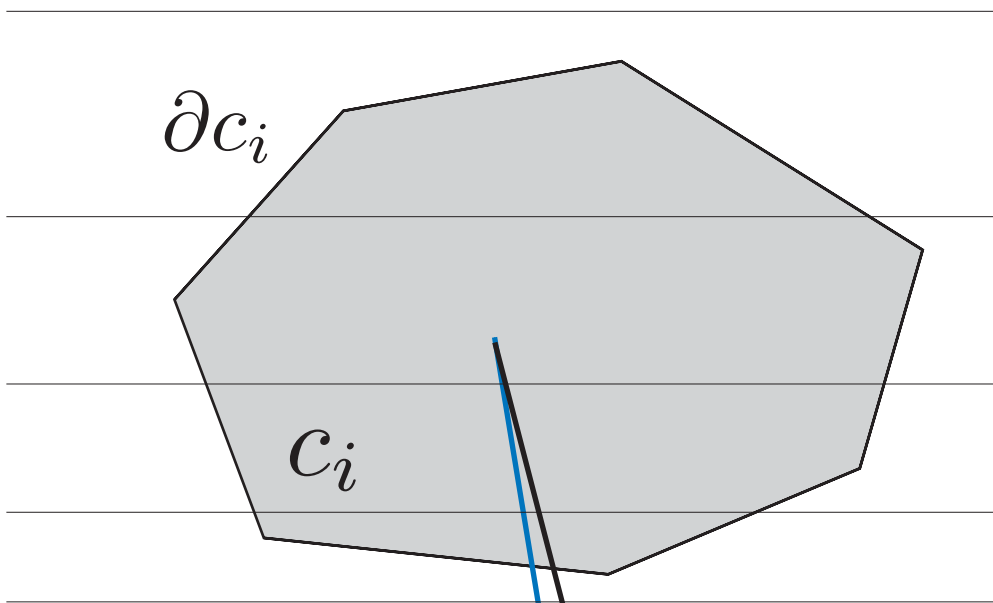
- Numerical truncation errors for 30 vertical layers, 2nd order scheme

$$(\delta w)_{\text{num.}} = O \left(\frac{1}{\mathbf{M}^2} \Delta x^2 \right) \sim \frac{(1/30)^2}{(0.03)^2} \sim O(1)$$

Origin of unbalanced truncation errors

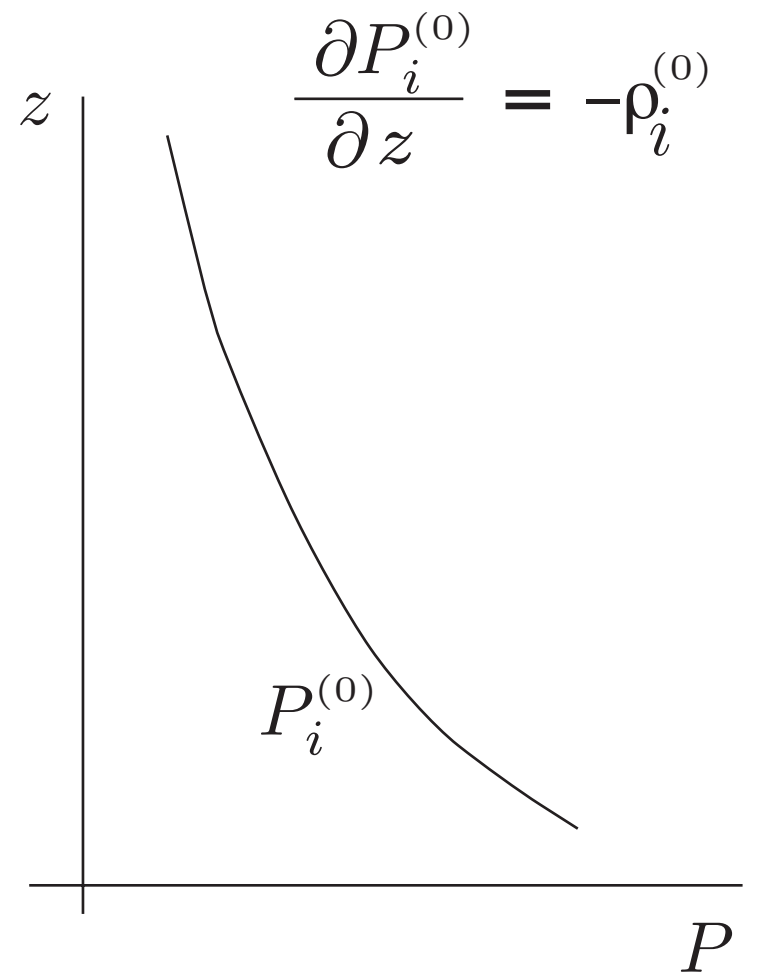


Archimedes' principle



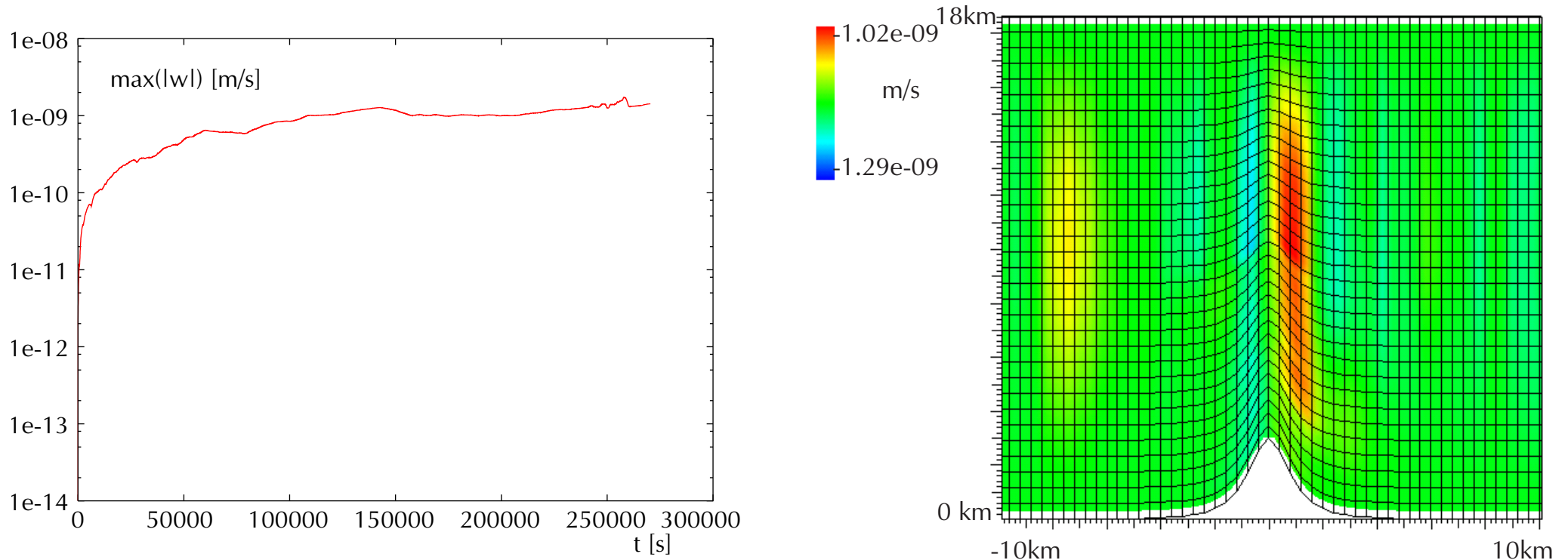
$$\frac{1}{|C_i|} \oint_{\partial C_i} P_i \mathbf{n} dS$$

$$\frac{1}{|C_i|} \oint_{\partial C_i} (P_i - P_i^{(0)}) \cdot \mathbf{n} dS$$



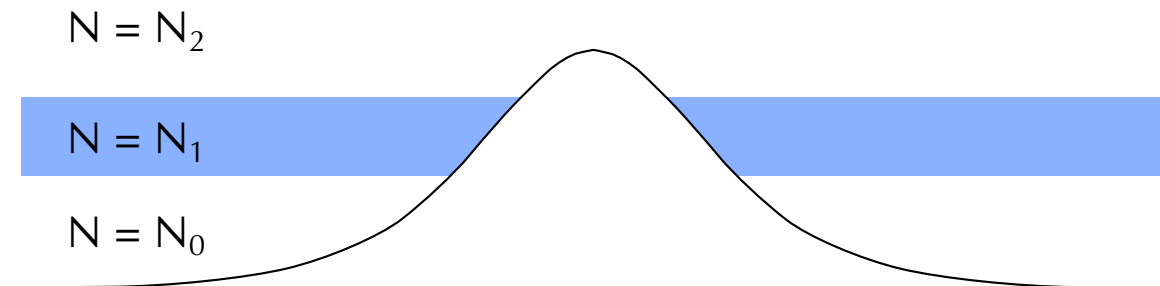
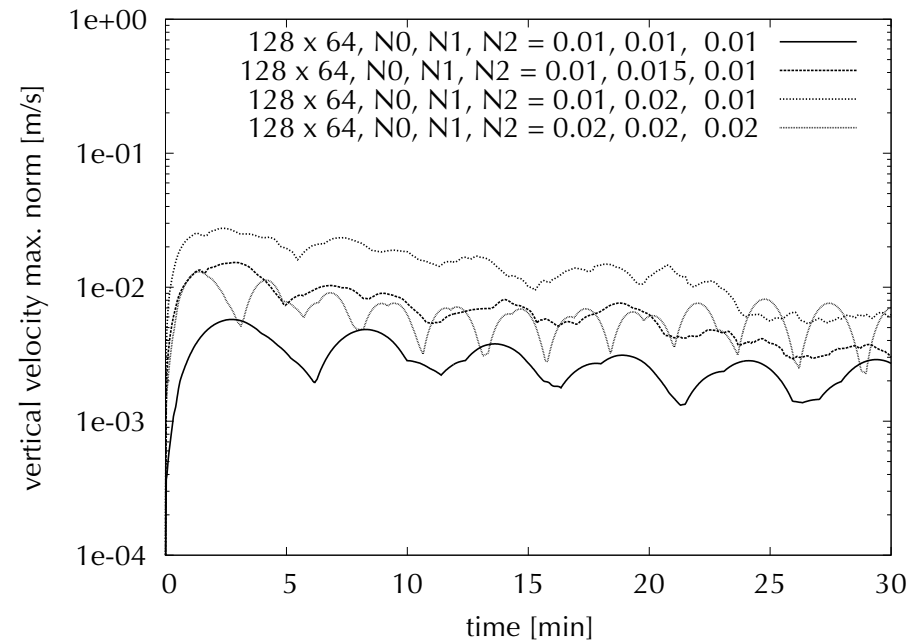
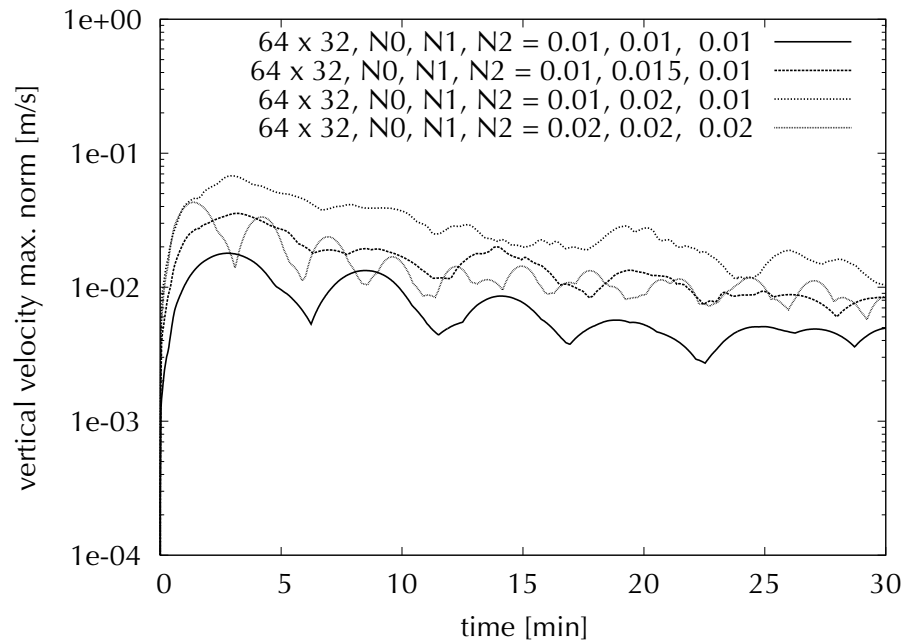
Archimedes' principle

Piecewise linear potential temperature

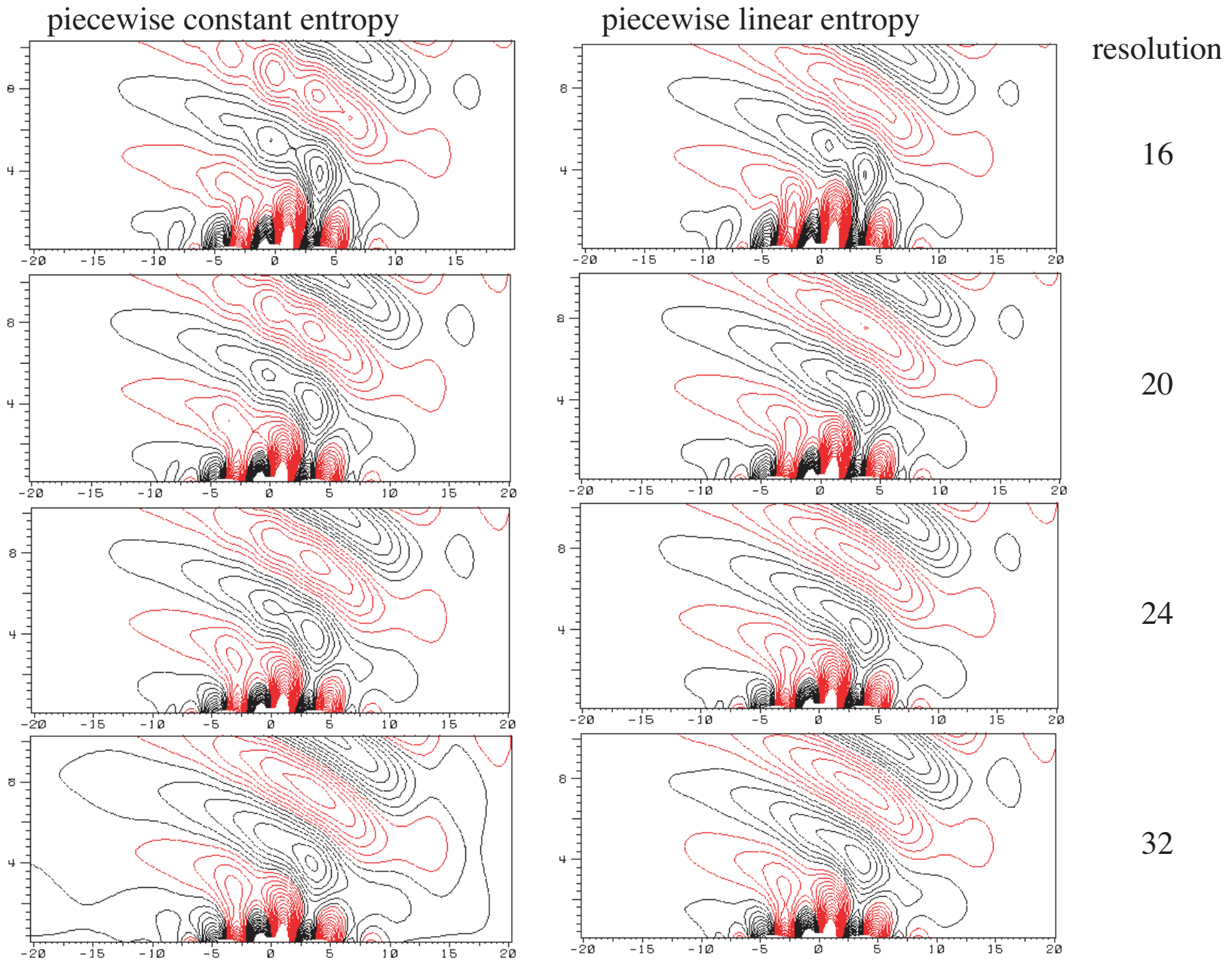


No accumulation of unbalanced truncation errors

Inversion layers at a mid-mountain level



No accumulation of unbalanced truncation errors



N. Botta, et al.: Well-Balanced Finite Volume Methods for Near-Hydrostatic Flows, JCP, (2004)

Convergence vs. marginal resolution (Schaer's Test)

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The Coriolis **source**-term

$$2 \boldsymbol{\Omega} \times \rho \mathbf{v}$$

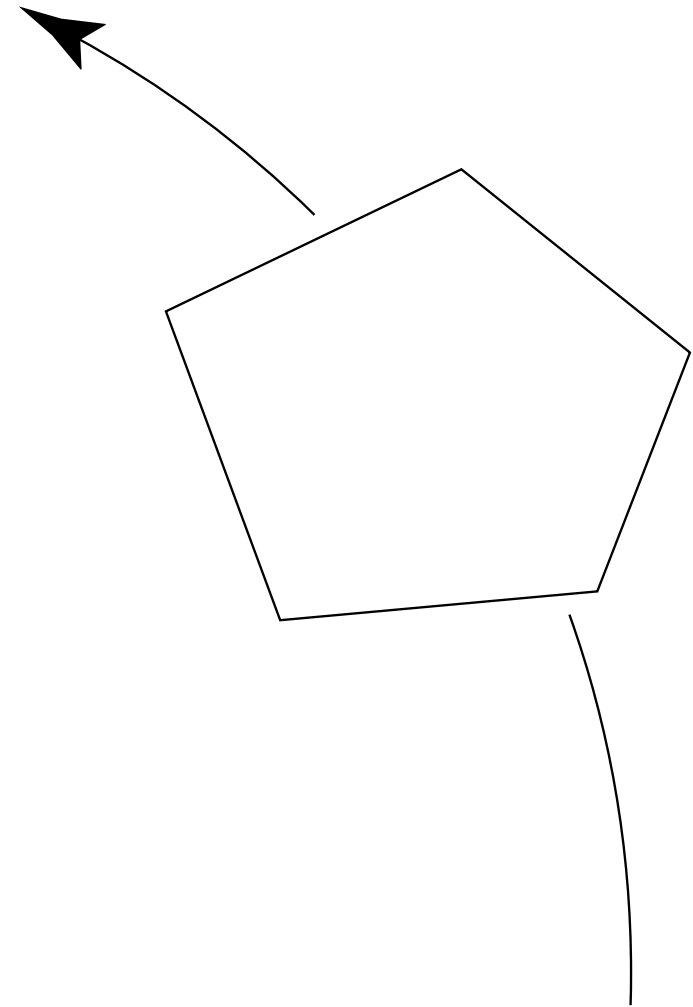
is the result of a coordinate transformation.

Standard numerical discretizations do not maintain momentum conservation.

Idea: Apply coordinate transformation to **conservative moving grid discretization** in the inertial frame.

$$2 \boldsymbol{\Omega} \times \rho \mathbf{v} \Rightarrow \text{rotationterm} + \text{fluxterm}$$

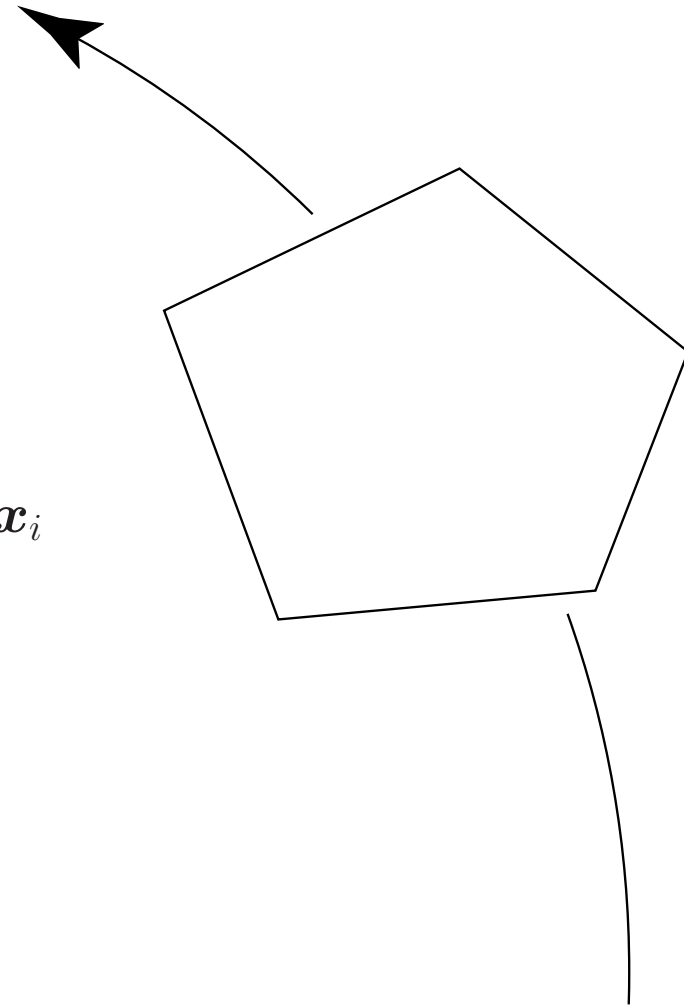
(work in progress with **E. Audusse**)

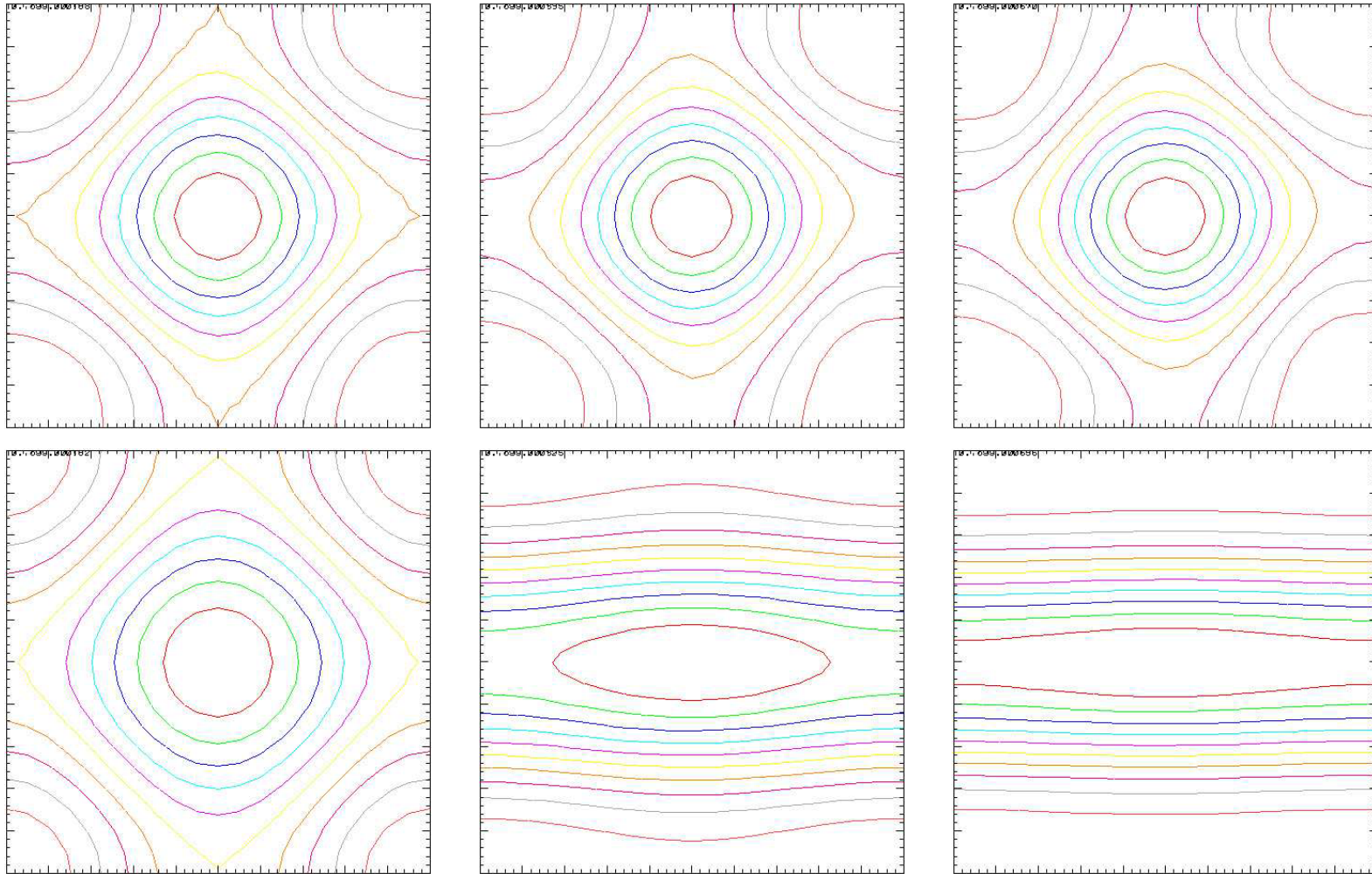


Shallow water version (1st order)

$$h_i^{n+1} = h_i^n - \frac{\Delta t}{|C_i|} \sum_j F_{ij}^h$$

$$\begin{aligned} \mathbf{q}_i^{n+1} = & \underline{R(\Delta t) \mathbf{q}_i^n} - h_i^{n+1} \boldsymbol{\Omega} \times \mathbf{x}_i + h_i^n R(\Delta t) \boldsymbol{\Omega} \times \mathbf{x}_i \\ & - \frac{\Delta t}{|C_i|} \sum_j \mathbf{F}_{ij}^q \\ & - \underline{\frac{\Delta t}{|C_i|} \sum_j (F_{ij}^h \boldsymbol{\Omega} \times \mathbf{x}_{ij})} \end{aligned}$$



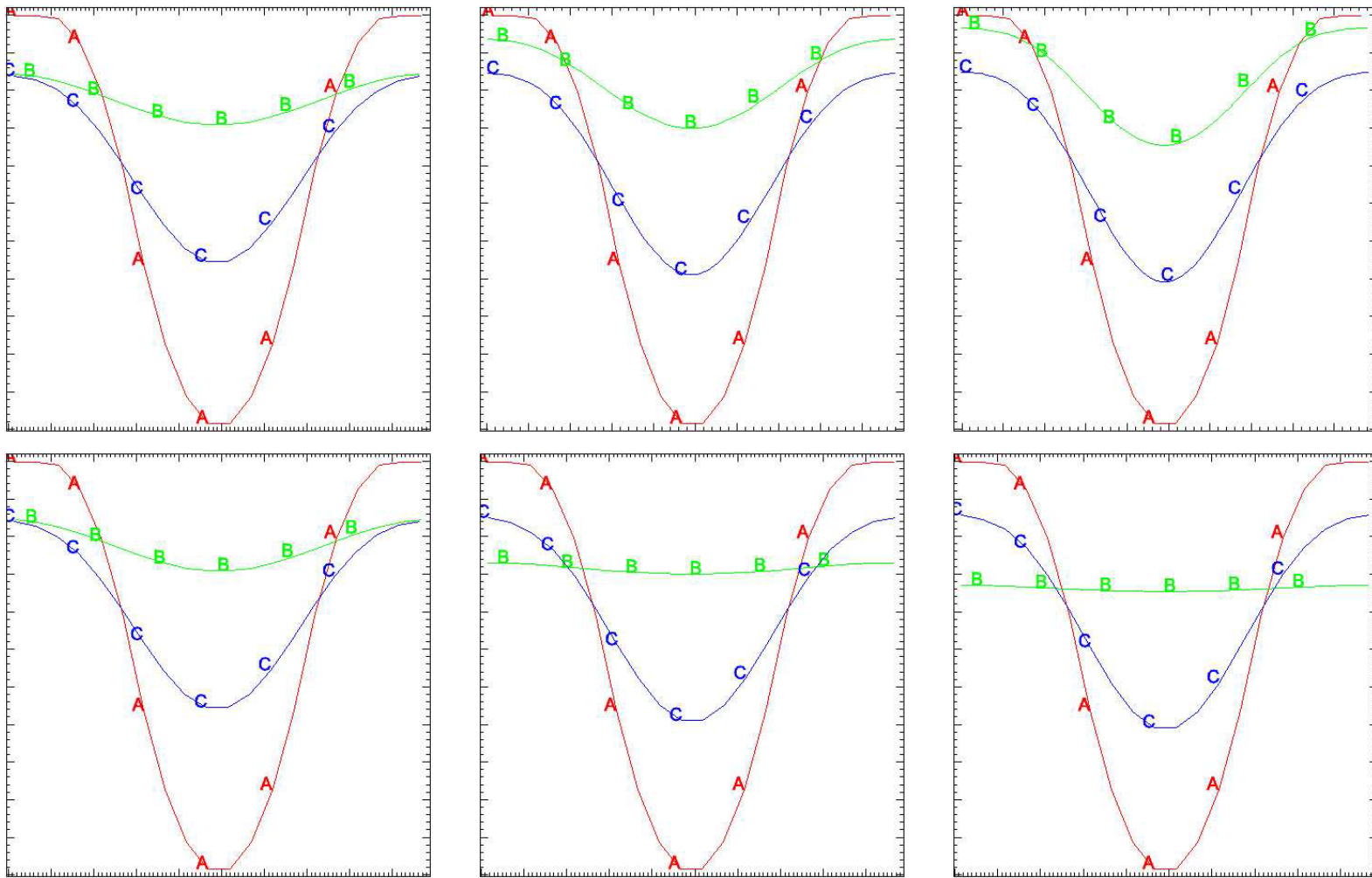


Water height for stationary vortex test:

Top: Cross-wind scheme, Bottom: Standard scheme

left to right: 20 cells horizontal / 20, 40, 80 cells in the vertical

Momentum conservation vs. Coriolis



Water height for stationary vortex test:

Top: Vertical cross section, Bottom: Horizontal cross section

left to right: 20 cells horizontal / 20, 40, 80 cells in the vertical

initial data, cross-wind scheme, standard scheme

Momentum conservation vs. Coriolis

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Three zero Mach number, anelastic regimes:

1. Inkompressible flow (no gravity)

2. Ogura-Phillips-Anelastic flow ($\theta = \mathbf{1} + \mathbf{M}^2\theta'$)

3. Buoyancy-controlled flow ($\theta = \Theta_0(z) + \mathbf{M}^2\theta'$)

Regime 1: Incompressible flow

Leading order result:

$$p = P_\infty + \varepsilon^4 p'$$

Consequences:

$$\rho e = \frac{p}{\gamma - 1} + \varepsilon^4 \frac{\rho \mathbf{v}^2}{2} \rightarrow \frac{P_\infty}{\gamma - 1} \quad (\varepsilon \rightarrow 0)$$

$$(\rho e)_t + \nabla \cdot ([\rho e + p] \mathbf{v}) = 0 \quad \rightarrow \quad \nabla \cdot \mathbf{v}^{(0)} = 0$$

$$\rho_t + \nabla \cdot (\rho \mathbf{v}) = 0 \quad \rightarrow \quad \rho_t^{(0)} + \mathbf{v}^{(0)} \cdot \nabla \rho^{(0)} = 0$$

$$(\rho \mathbf{v})_t + \nabla \cdot (\rho \mathbf{v} \circ \mathbf{v}) + \frac{1}{\varepsilon^4} \nabla p = 0 \quad \rightarrow \quad (\rho \mathbf{v})_t^{(0)} + \nabla \cdot (\rho \mathbf{v} \circ \mathbf{v})^{(0)} + \nabla p' = 0$$

Task:

Construct a scheme for low Mach number flows, which ...

- conserves mass, momentum, and total energy,
- is at least second order accurate in space and time,
- requires the solution of at most linear, scalar equations (possibly a few of them per time step)
- reduces, for $\mathbf{M} = 0$, to the variable density, incompressible solver in Schneider et al., JCP, (1999)
- reduces, for $\mathbf{M} = 1$, to a second order Godunov-type scheme

Explicit / **implicit** split for the Euler equations

(R.K., JCP (1995); M. Minion & R.K, (2005))

$$\rho_t + \nabla \cdot (\rho \mathbf{v}) = 0$$

$$(\rho \mathbf{v})_t + \nabla \cdot (\rho \mathbf{v} \circ \mathbf{v}) + \nabla p + \underbrace{\left(\frac{1}{\mathbf{M}^2} - 1 \right) \nabla \mathbf{p}_*}_{\text{blue underline}} = 0$$

$$(\rho e)_t + \nabla \cdot ([\rho e + \boldsymbol{\pi}] \mathbf{v}) = 0$$

$$\rho e = \frac{p}{\gamma - 1} + \mathbf{M}^2 \frac{\rho \mathbf{v}^2}{2}$$

$$\boldsymbol{\pi} = \underbrace{\left(1 - \mathbf{M}^2 \right) \mathbf{p}_*}_{\text{blue underline}} + \mathbf{M}^2 p$$

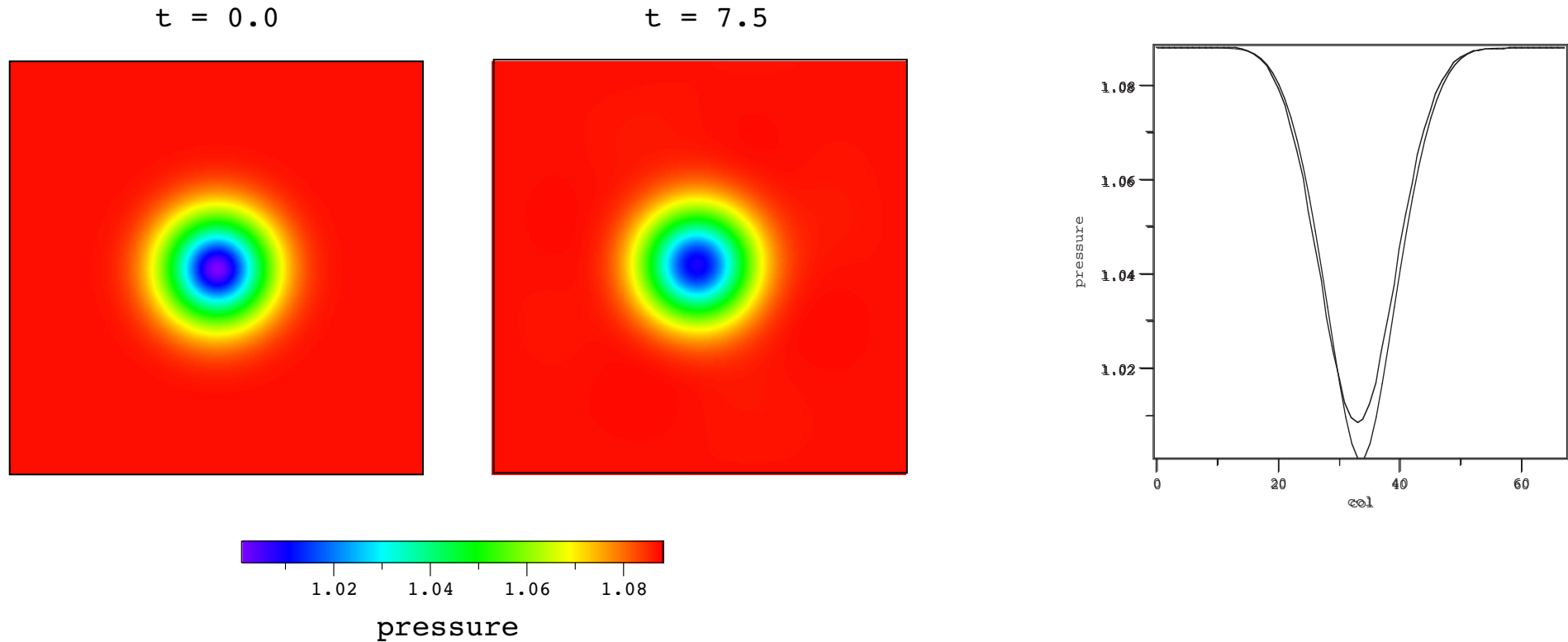
Energy balance (Divergence correction for energy fluxes)

$$\begin{aligned} \mathbf{M}^2 \left(\frac{1}{\Delta t} \left[\frac{\delta \mathbf{p}_*'}{\gamma - 1} + \left(\frac{\rho \mathbf{v}^2}{2} \right) \Big|_n^{**} - \mathbf{v}^{**} \cdot \tilde{\nabla} \delta \mathbf{p}_*' \right] + (1 - \mathbf{M}^2) \tilde{\nabla} \cdot (\delta \mathbf{p}_*' \mathbf{v}^{**}) \right) \\ = - \left(\tilde{\nabla} \cdot (\rho h^* \mathbf{v}^{**}) - \frac{\Delta t}{2} \tilde{\nabla} \cdot (h^* \tilde{\nabla} \delta p) \right) \end{aligned}$$

$\mathbf{M} = 0 \Rightarrow$ classical **linear** projection step

$\mathbf{M} \ll 1 \Rightarrow$ **linear**, non-singular inclusion of weak compressibility

Advected homentropic vortex at $M = 0.2$



Convergence rates: $v \rightarrow h^2$, $p \rightarrow h^{1.5}$, $\rho \rightarrow h^1$

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Second projection: discrete approximations of $\nabla \cdot \left(\frac{1}{\rho} \nabla p \right) = r$

Finite elements (with test functions ϕ_{ν}):

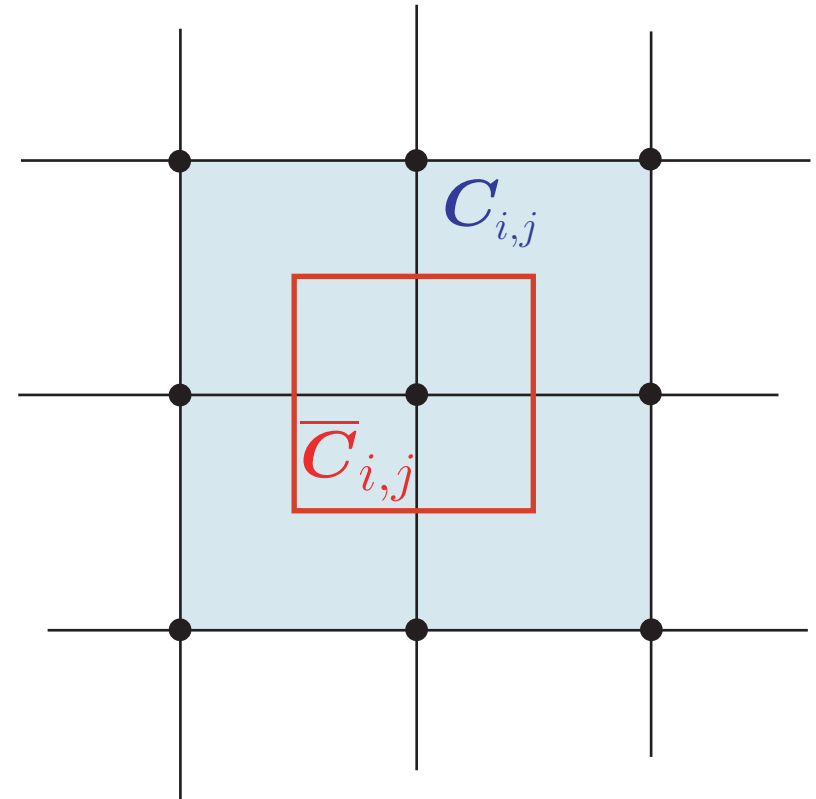
$$\forall \nu : \int_{\bar{C}_{i,j}} \left[\frac{1}{\rho} \nabla p \cdot \nabla \phi_{\nu} - r \phi_{\nu} \right] dx = 0$$

where

$$p(x, y) = \sum_{\mu} \hat{p}_{\mu} \phi_{\mu}$$

Finite volumes on dual cells

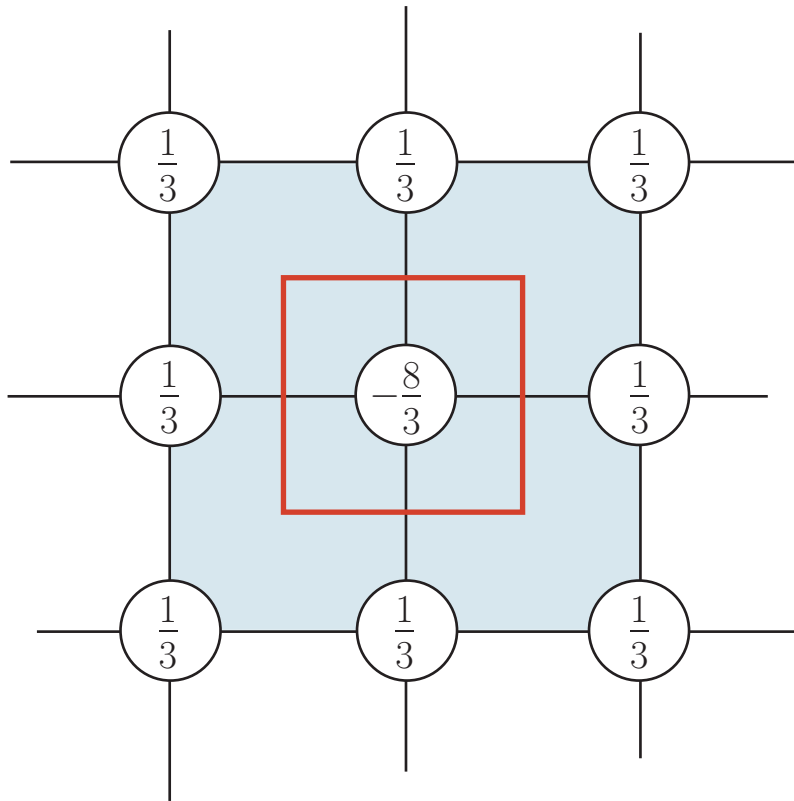
$$\oint_{\partial \bar{C}_{i,j}} \frac{1}{\rho} \nabla p \cdot \mathbf{n} \, d\sigma - \int_{\bar{C}_{i,j}} r \, dx = 0$$



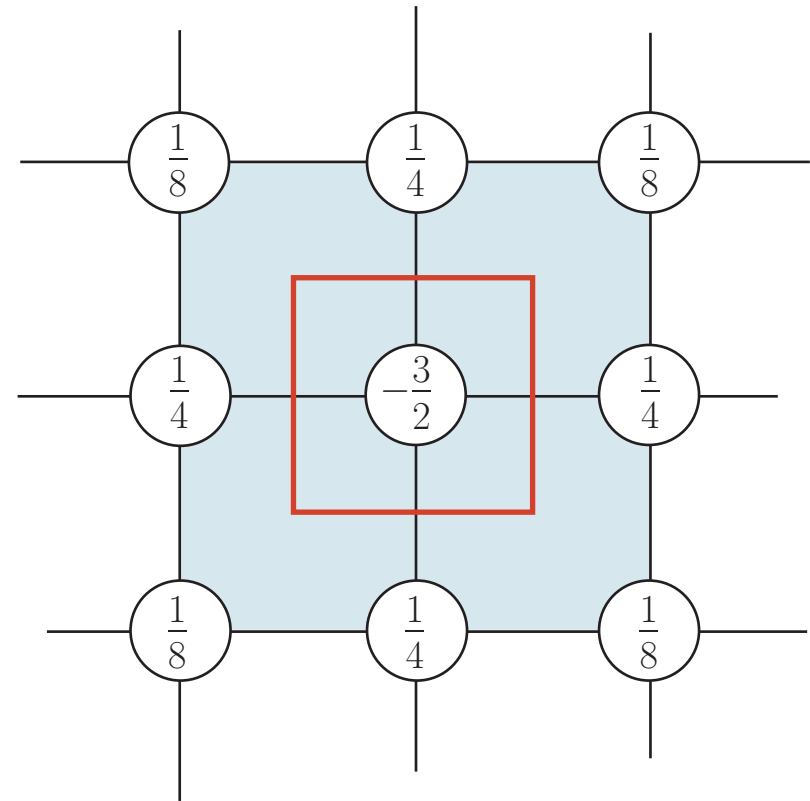
Bilinear pressure on FV-cells $\bar{C}_{i,j}$

Bilinear pressure distributions: FE vs. FV stencil

Example: (robust, second order)* stencils for $\Delta p = \nabla \cdot \nabla p$



Conforming **finite elements**

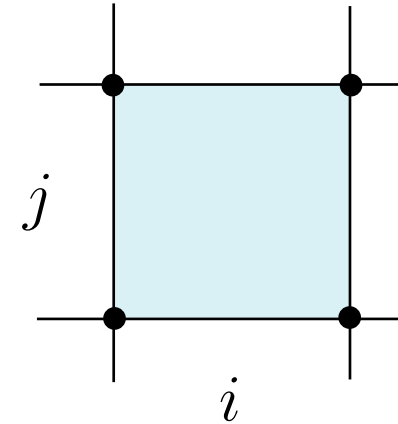


Finite volumes on dual cells:

* E. Süli: SIAM J. Num. Analysis, 28(5):1419-1430, 1991.

Cell averaged updates

$$(\rho \mathbf{v})_{i,j}^{n+1} = (\rho \mathbf{v})_{i,j}^* - \Delta t \begin{pmatrix} (\partial_x p^{(2)})_{i,j} \\ (\partial_y p^{(2)})_{i,j} \end{pmatrix}$$



conservative gradient approximation, e.g.,

$$\left(\partial_x p^{(2)}\right)_{i,j} \approx \frac{1}{2\Delta x} \left(\left[p_{i+\frac{1}{2},j+\frac{1}{2}} + p_{i+\frac{1}{2},j-\frac{1}{2}} \right] - \left[p_{i-\frac{1}{2},j+\frac{1}{2}} + p_{i-\frac{1}{2},j-\frac{1}{2}} \right] \right)$$

Bilinear pressure distributions: FV momentum updates

Local updates within cells $C_{i,j}$:

$$(\rho \mathbf{v})^{n+1}(x, y) = (\rho \mathbf{v})^*(x, y) - \Delta t \nabla p^{(2)}(x, y)$$

where

$$\nabla p^{(2)}(x, y) \Big|_{i,j} = \begin{pmatrix} (\partial_x p^{(2)})_{i,j} + (y - y_j) \left(\partial_{xy}^2 p^{(2)} \right)_{i,j} \\ (\partial_y p^{(2)})_{i,j} + (x - x_i) \left(\partial_{xy}^2 p^{(2)} \right)_{i,j} \end{pmatrix} .$$

Second projection modifies **piecewise linear reconstruction**:

$$\begin{aligned} \left(\partial_x(\rho v) \right)_{i,j}^{n+1} &= (\partial_x(\rho v))_{i,j}^* - \Delta t \left(\partial_{xy}^2 p^{(2)} \right)_{i,j} \\ \left(\partial_y(\rho u) \right)_{i,j}^{n+1} &= (\partial_y(\rho u))_{i,j}^* - \Delta t \left(\partial_{xy}^2 p^{(2)} \right)_{i,j} \end{aligned}$$

Bilinear pressure distributions: FV momentum updates

Exact projection achieved with:

$$(\nabla \cdot \mathbf{u})_{i,j}^{n+1} = \frac{1}{|\partial \bar{C}_{i,j}|} \oint_{\partial \bar{C}_{i,j}} \left(\frac{1}{\rho} \mathbf{m} \right)^{n+1} \cdot \mathbf{n} \, d\sigma$$

where

$$\mathbf{m}^{n+1}(x, y) \Big|_{i,j} = \begin{pmatrix} (\rho u)_{i,j}^{n+1} + (x - x_i) (\partial_x(\rho u))_{i,j}^* + (y - y_j) \left(\partial_y(\rho u) \right)_{i,j}^{n+1} \\ (\rho v)_{i,j}^{n+1} + (x - x_i) \left(\partial_x(\rho v) \right)_{i,j}^{n+1} + (y - y_j) (\partial_y(\rho v))_{i,j}^* \end{pmatrix}$$

Bilinear pressure distributions: Exact projection

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Regime 3: Buoyancy-controlled anelastic flow $\theta = \Theta_0(z) + \mathbf{M}^2 \theta'$

Leading order results:

$$p = P_0(z) + \mathbf{M}^2 p' \quad \rho = R_0(z) + \mathbf{M}^2 \rho' \quad R_0(z) = \Theta_0(z) P_0(z)^{1/\gamma}$$

Consequences:

$$\rho e = \frac{p}{\gamma - 1} + \mathbf{M}^2 \frac{\rho \mathbf{v}^2}{2} + \rho g z \rightarrow \frac{P_0(z)}{\gamma - 1} + R_0(z) g z \quad (\mathbf{M} \rightarrow 0)$$

$$\begin{aligned} (\rho e)_t + \nabla \cdot ([\rho e + p] \mathbf{v}) = 0 &\rightarrow \nabla \cdot \left(\mathbf{v}^{(0)} R_0(z) \left[\frac{\gamma}{\gamma - 1} \frac{P_0}{R_0}(z) + g z \right] \right) = 0 \\ \rho_t + \nabla \cdot (\rho \mathbf{v}) = 0 &\rightarrow \nabla \cdot \left(\mathbf{v}^{(0)} R_0(z) \right) = 0 \end{aligned}$$

Unfortunately

$$\left[\frac{\gamma}{\gamma - 1} \frac{P_0}{R_0} + g z \right] \equiv \Phi(z) \neq \text{const.} \Rightarrow \nabla_{\parallel} \cdot \mathbf{u}^{(0)} = 0, \quad \text{and} \quad w^{(0)} = 0$$

Anelastic limits

Regime 3: Buoyancy-controlled anelastic flow $\theta = \Theta_0(z) + \mathbf{M}^2\theta'$

Leading order results:

$$p = P_0(z) + \mathbf{M}^2 p' \quad \rho = R_0(z) + \mathbf{M}^2 \rho' \quad R_0(z) = \Theta_0(z) P_0(z)^{1/\gamma}$$

Consequences (cont'd):

$$(\rho \mathbf{v})_t + \nabla \cdot (\rho \mathbf{v} \circ \mathbf{v}) + \frac{1}{\mathbf{M}^2} (\nabla p + \rho g) = 0 \quad \rightarrow \quad (\rho \mathbf{v})_t^{(0)} + \nabla \cdot (\rho \mathbf{v} \circ \mathbf{v})^{(0)} + (\nabla p' + \rho') = 0$$

$(P_0, \Theta_0, p', \rho')$ must be maintained as primary variables in a numerical implementation

Generalization to anelastic regimes:

Energy balance (**Divergence correction for energy fluxes**)

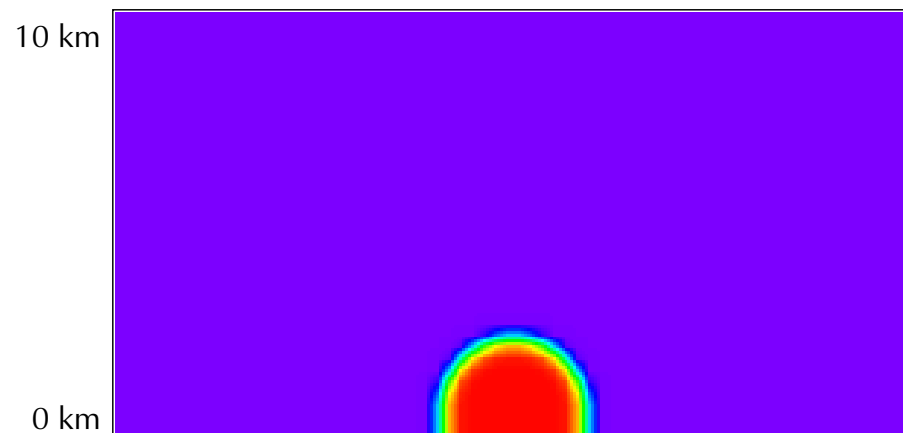
$$\begin{aligned} \mathbf{M}^2 \left(\frac{1}{\Delta t} \left[\frac{\delta \mathbf{p}_*'}{\gamma - 1} + \left(\frac{\rho v^2}{2} \right) \Big|_n^{**} - \Delta t \mathbf{v}^{**} \cdot \tilde{\nabla} \delta \mathbf{p}_*' \right] + (1 - \mathbf{M}^2) \tilde{\nabla} \cdot (\delta \mathbf{p}_*' \mathbf{v}^{**}) \right) \\ = - \left(\tilde{\nabla} \cdot (\rho h^* \mathbf{v}^{**}) - \frac{\Delta t}{2} \tilde{\nabla} \cdot (h^* [\tilde{\nabla} \delta \mathbf{p}_*' + \delta \rho_*' \mathbf{k}]) \right) \end{aligned}$$

Mass balance (**Divergence correction for mass fluxes**)

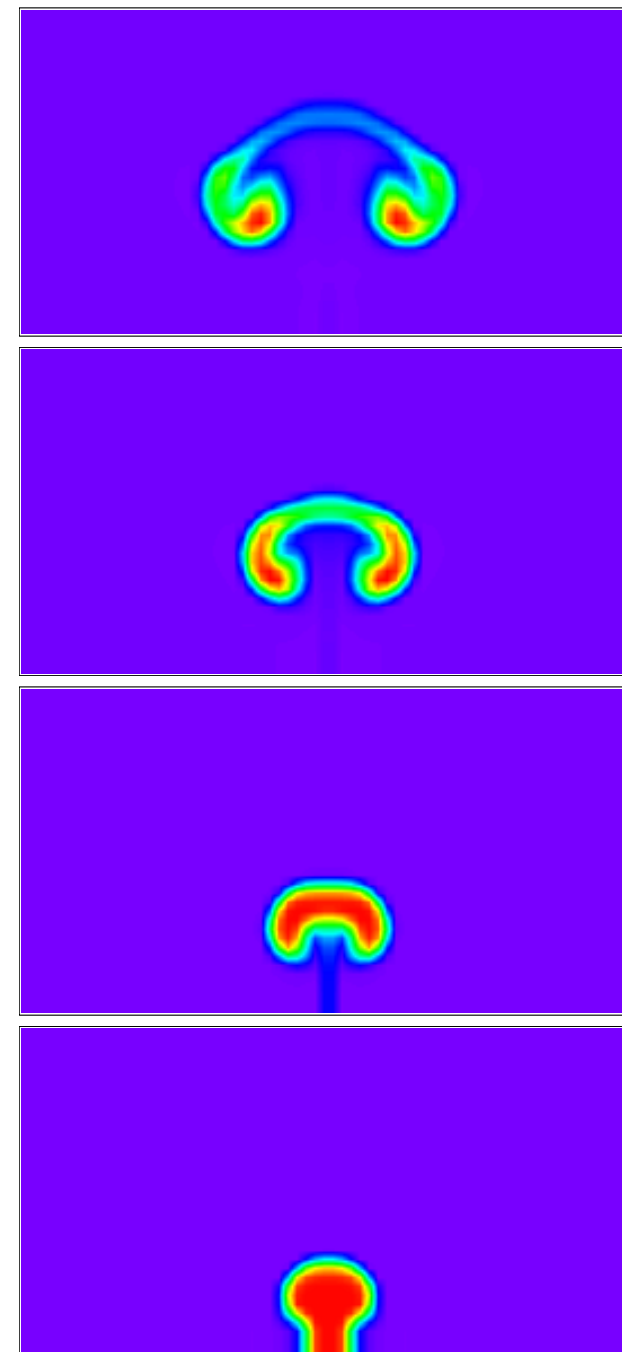
$$\frac{\mathbf{M}^2}{\Delta t} \delta \rho_*' = - \left(\tilde{\nabla} \cdot (\rho^* \mathbf{v}^{**}) - \frac{\Delta t}{2} \tilde{\nabla} \cdot ([\tilde{\nabla} \delta \mathbf{p}_*' + \delta \rho_*' \mathbf{k}]) \right)$$

Potential Temperature:

$$\theta = \frac{p^{\frac{1}{\gamma}}}{\rho}$$



Rising thermal plume ($\delta\theta/\theta = 0.1$) in a 10 km neutral atmosphere



An example

Conclusions

