Clear Air Turbulence (CAT) Simulation and Observation: Implications for Parameterization

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Observations of CAT layers with mu-rass in Japan



Example of turbulent layers observed with mu-rass in Japan. Layers range from 100-150m (resolution limit) up to 1km. Some layers persist over 4-day observing period and are seen to be rejuvenated on a diurnal time scale.

High-Resolution Radar Backscatter

Ierkic, Woodman & Perillat, Radio Science 25, 941 (1990)



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INVVKA/CORA



Estes Park, Colorado, 1979 (photo by Bob Perney)



Colorado Springs, Colorado, 2000 (photo by Tye Parzybok)

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Denver, Colorado, 1953 (photo by Paul E. Branstine)



Lafayette, Colorado, 2002 (photo by Joe Werne)

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Noctilucent Clouds, Kustavi, Finland, 1989 (photo by Pekka Parviainen)

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Modeling challenges for stably stratified clear air turbulence:

- Turbulence is episodic and spatially confined
- Fossil events can serve to precondition future events
- Depths of most isolated layers are subgrid scale
- Mixing can be non-local, with gravity waves providing remote subgrid-scale momentum transfer



Noctilucent Clouds, Kustavi, Finland, 1989 (photo by Pekka Parviainen)

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CAT layers: What are their signatures?



CAT layers: How typical or rare are they?

Coulman, Vernin & Fuchs, Applied Optics 34 5461 (1995)





CAT layers: What are their characteristics?

Data from 350 balloon profiles from VTMX, Salt Lake City, Utah, Oct 2000. Identical results obtained with CASES-99 data, Kansas, Oct 1999, despite much flatter terrain.



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Unresolved CAT: Is It Important?

1992 CAT Event

- 9 Dec 1992 Front Range windstorm, Evergreen, CO
- DC-8 Cargo aircraft encounters two full minutes of intense turbulence
- Left engine and 12-feet of wing ripped from plane
- Pilot landed safely at Stapleton



Clark, et al., J. Atmos. Sci. <u>57</u> 1105-1131 (2000)







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Unresolved CAT: Is It Important?

50 years of U2 flights

- U2 reconnaissance 50 years of high-altitude experience
- Many aborted missions due to high-altitude wave activity
- Damaged / destroyed aircraft
- One pilot killed





1997 CAT Event

- 29 Dec 1997, United flight 826 to Honolulu from Japan
- Boeing 747 encountered severe turbulence at 33,000 ft
- Dropped 1000 ft, 110 passengers injured, 1 killed

The New York Times

December 29, 1997 Jet Hits Turbulence; 110 Hurt and a Woman Dies

A United Airlines jumbo jetliner with 393 people aboard hit severe air turbulence over the Pacific Ocean on Sunday night, killing one Japanese woman and injuring 110 other passengers.

Passengers and serving carts were flung to the ceiling as the plane dived 1,000 feet when it flew into the turbulence at 33,000 feet, officials said.

The plane, flight 826 bound for Honolulu with 374 passengers and 19 crew members, flew back safely to Narita, Tokyo's main international airport, and landed at 2:25 A.M.

The officials said a 32-year-old Japanese woman died and 10 passengers were injured seriously enough to remain in hospitals.

The dead passenger, who was not immediately identified by airline officials, was flung to the ceiling and died shortly after she was taken to a hospital, officials said.

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Unresolved CAT: Is It Important?

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1997 CAT Event

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What can we do about simulating/modeling?

Can we anticipate intense events without increasing computational cost?

What about (slightly) smaller-scale events?

Probabilistic modeling (Poulos & Burns 2003)

and a Woman Dies

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Wind Shear Simulation





$$Ri = \frac{N^2 h^2}{U_0^2} \quad Re = \frac{U_0 h}{v} \quad Pe = \frac{U_0 h}{\kappa}$$

- Stream-function/vorticity formulation of the Boussinesq equations
- Fully spectral, 3D FFT's = 80% cost
- Radix 2,3,4,5 FFTs
- Spectral modes and NCPUs must be commensurate
- Communication: shmem and MPI, global transpose (all-to-all), data reduction
- Parallel I/O every ~ 60 δt
- Up to 4000 x 2000 x 2000 modes
- 6 Grand Challenge and 2 DoD CAP awards
- Logged over 10 million CPU hours; generated over 300 Tbytes of archived data

$$\partial_t u + \omega \times u = Re^{-1}\nabla^2 u - \nabla P + Ri\,\Theta\,\hat{z}$$

 $\partial_t \theta + u \cdot \nabla \theta = P e^{-1} \nabla^2 \theta$

$$\nabla \cdot u = 0$$

Wind Shear Simulation





 $\nabla \cdot u = 0$

Hi-Res Wind-Shear Simulations



















Hi-Res Wind-Shear Simulations

Extensive validation using balloon, radar, and aircraft data resulted in:

- Agreement with flow morphology deduced from cloud imagery, previous work
- Measured atmospheric mean profiles for wind, temperature, Cn², and Ri
- Structure-function scaling ~ 2/3 and 2/5, consistent with later aircraft measurements.
- Turbulence inner scale $\ell_0 = 7.4 \ \ell_K$, consistent with tower data.
- 2nd-order structure functions for T and U consistent with atmospheric BL measurements: $C_T^2 = 3.3 \ \epsilon^{-1/3} \chi$, $C_U^2 = 2.1 \ \epsilon^{2/3}$
- 2nd-order structure function ratios $C_V^2/C_U^2 = 1.06$, $C_W^2/C_U^2 = 0.6$, consistent with later aircraft data (and both far from the predicted value of 1.33).
- Dynamic SGS LES with gradient-diffusion model cannot be validated due to small box size. Need larger domain to make LES progress.

DoD CAP Wind-Shear Simulations





DoD CAP Wind-Shear Simulations







Ri=0.20 Re=4000

Vortex tubes viewed from above, Ri=0.05



Vortex tubes viewed from above, Ri=0.20



t=82 : turbulence reaches braids

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t=54 : secondary instability





t=68 : KE, PE local minima

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t=85 : KE, PE secondary maxima

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t=111 : turbulence intensity and vorticity maxima

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t=37 : maximum laminar amplitude





t=54 : maximum billow amplitude, turbulence erupts in billow cores





t=66 : PE peaks





t=82 : turbulence reaches braids

KE and $max(\omega)$ evolution



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Grob G520T Egrett (Airborne Research Australia)

- ★ Altitude: up to 15 km
- ★ Airspeed: 100 m/s
- ★ Endurance: 8 hrs
- ★ 3 NOAA BAT probes (under wings and high on tail)
- ★ T and (U,V,W) at 50 Hz (2 m horizontal resolution)









Simulation Results Ri=0.05
Aircraft Data (Wales)

Observed Cliff-Ramp structures are much more consistent with the High-Ri simulation results:
Image: Construction of the text of tex of text of text of text of







0.6

0.4

0.0

-0.6 -0.8

0.2

0.0

-0.2

(HB)

Simulation Results Ri=0.05 Aircraft Data (Wales) 0.6 Observed Cliff-Ramp structures are much more 0.4 0.2 (Hg) consistent with the High-Ri simulation results: 0.0 -0.6 -0.8 T-cliffs are consistently sharp 1. 0.2 U rises sharply at T-cliff 2. 0.0 (°n) n W fluctuations are suppressed in braid region 3. -0.2 No observed large-scale variation in W at 4. -0.4most-unstable KH wavelength 0 Why the inconsistency? V (00) 0 1. KH clouds nearly always appear as deep, round, low-Ri billows, -02. while cliff-ramp structures in the aircraft measurements are consistent only with high-Ri flows. 0 0 00 (°n) M 0 2. The most obvious feature of every data set is reported first. -01. Clouds are too large to visualize shallow high-Ri layers. 82

Despite dramatic differences with Ri, universal features exist.



Final N/S=1.05 (Ri=0.55)



Finding our way in the terra incognita

Yesterday John Wyngaard described the *terra incognita* between mesoscale-model and LES-model resolutions.

• Prof. Wyngaard proposed SGSmodel improvements to address this middle land for model resolution.

φ(κ)

- Additional improvements will likely be required as well.
- Partially resolved dynamics require sophisticated SGS models that can anticipate motions in the energy-containing portion of the spectrum and represent their effects.
- Propagating waves, overturning wind-shear, and wave/mean-flow interactions at or below the model cut-off scale can be important.



FIG. 1. A schematic of the turbulence spectrum $\phi(\kappa)$ in the horizontal plane as a function of the horizontal wavenumber magnitude κ . Its peak is at $\kappa \sim 1/l$, with *l* the length scale of the energetic eddies; Δ is the scale of the smoothing filter. In the mesoscale limit (left), $\Delta_{\text{meso}} \gg l$ and none of the turbulence is resolved. In the LES limit (right), $\Delta_{\text{LES}} \ll l$ and the energy-containing turbulence is resolved.

Finding our way in the terra incognita

In order to capture the effects of unresolved or partially resolved dynamics, SGS models must be knowledgeable of the dynamical motions at or near (including below) the model filter scale.

Important unresolved processes must be included in a probabilistic manner to ensure they are represented by the model.

Examples of stochastic forcing used near the cut-off length scale to include the effects of unresolved turbulent processes include:

1. The current ECMWF ensemble forecast model.

2. 'Stochastic backscatter' in LES studies of the atmospheric boundary layer.

To address the *terra incognita*, we are investigating using existing DNS solutions combined with observations to improve current stochastic physics and/or stochastic backscatter methods by specifying more defensible probability distributions.

Mason & Thomson, "Stochastic backscatter in LES of boundary layers." JFM, 242, 51-78 (1992)

Westbury, Dunn & Morrison, "Analysis of a stochastic backscatter model for the LES of wall-bounded flow." Europ. J. of Mech. B/Fluids, 23, 737-758 (2004).

Palmer et al., "Representing model uncertainty in weather and climate prediction," Ann. Rev. Earth Planet. Sci. 33, 163-93 (2005)

A = quantity you want to predict, e.g., τ_{ij} , f_i , G_i , $\Delta_{\vec{r}}T^2$, ... F = NWP forecast variables, e.g., T, P, U, V, W, ...Y_i = important unknowns, e.g., L_i , ΔU_i , ΔT_i , $Ri_i^{(o)}$, a_i , Re_i , z_i , ...

atmospheric turbulence patch parameters (layer depth, velocity and temperature jump, local Richardson number, age, Reynolds number, altitude, ...)

prediction takes form of a joint probability distribution:

$$[A, F] = \int [A, F, Y] dY$$

where Y represents a collection of layers: $[Y] = [Y_1, Y_2, ..., Y_N]$

notation:

[a] - probability of 'a' [a,b]- joint probability of 'a' and 'b'

[a | b] - conditional probability of 'a' given 'b'

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Some layers Y will be resolved and some will have to be modeled. Therefore, we split the integral at $L_i = \Delta$, where Δ is the NWP model cut-off length scale.













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Conclusions

1. Shear and gravity-wave dynamics produce layered turbulence that is episodic in time.

2. The unresolved dynamics are important and their likelihood of occurrence must be modeled.

3. Measurements demonstrate infrequent but significant deep CAT events and numerous less important shallow CAT events that produce near-universal PDFs that can be characterized by two parameters (mean and variance).

4. DNS results for turbulent shear have been validated with balloon, radar, and aircraft data, and clear deviations from isotropic theory are seen for both the DNS and the observations.

5. Different morphology and evolution exists for weakly and strongly stratified events, permitting identification of the initial Ri in aircraft data.

6. Gravity-wave-breaking simulations demonstrate amplitude reductions from 1.1 to 0.3, far in excess of conventional linear saturation theory. Expect different degree of amplitude reduction for different frequency waves.

7. We have defined a Bayesian framework for probabilistic SGS formulations.

Ongoing Work

- 1. Constructing atmospheric Ri census via refined simulation/aircraft-data comparisons.
- 2. Developing needed BHM conditional PDFs from DNS results.

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